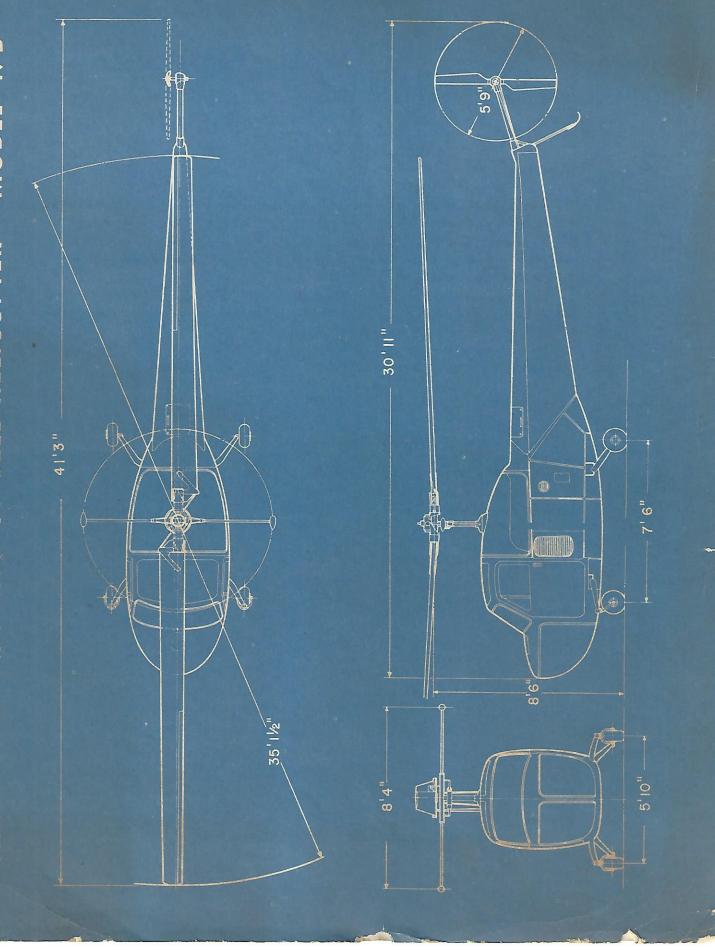
THREE-VIEW DRAWING OF BELL HELICOPTER . MODEL 47B





CUSTOMER'S SPECIFICATIONS FOR MODEL 47B

GENERAL DESCRIPTION

The Bell Model 47B is a two-place, single-engine helicopter featuring the Bell-patented two-blade rotor and an exclusive stabilizer bar. A small tail rotor compensates torque, offers precise directional control.

Comfortable seats are placed side by side within a two-door cabin forward of the engine. Large glassed-in areas assure both pilot and passenger unusual visibility. Abaft the engine section, there is a generous luggage compartment.

Standard equipment includes complete dual controls. For maximum convenience, a main rotor pitch control lever is mounted at the left of each seat. Otherwise, all instruments and engine controls are placed on a central column readily accessible to either occupant.

On the ground, the helicopter sits level on a quadracycle landing gear. Front wheels turn a full 360° and greatly facilitate ground handling.

PERFORMANCE DATA

The performance figures listed below are those for operation with the maximum gross load, in still air (0 miles per hour) under NACA standard atmospheric conditions. With normal wind velocity and average temperatures, increased performance is available.

High speed at sea-level	92 mph
Maximum rate of climb at sea-level	
Vertical rate of climb at sea-level	
Service ceiling	
Hovering ceiling (in ground effect)	3800 ft.

Time to climb to 5000 feet.	6.9 min.
Operating speed, sea-level, 75% power	er85 mph
Range, sea-level, 75% power	212 miles
Endurance, sea-level, 75% power	2.5 hrs.
Maximum endurance, 60% power	3.0 hrs.

WEIGHT

Gross weight (normal load)	2200 lbs.
Empty weight	1523 lbs.
Useful load	677 lbs.
Pilot	170 lbs.
Payload	290 lbs.
Fuel (32 gals. plus 1/4 gal. trapped)	193 lbs.
Oil (2 gals. plus 1.16 gal. trapped)	24 lbs.

STRUCTURE AND DESIGN

DIMENSIONS - MAIN ROTOR

Diameter 35.125 ft	Disc area	965 sq. ft.
Number of blades	Disc loading (normal gross weight) 2.2	28 lbs./sq. ft.
Blade area (total) 35.34 sq. ft	Power loading (normal gross weight). I	2.35 lbs./hp.

STABILIZER BAR

The stabilizer bar is a Bell exclusive. It is responsible in large measure for the Bell Helicopter's unusual flight steadiness at all speeds and altitudes.

This bar is mounted on a seesaw pivot immediately beneath the main rotor. It rotates with the mast . . . at right angles to rotor blades. However, because it is a large rotating weight, the bar tends to hold a given plane of rotation independent of either mast or rotor.

Linked mechanically to the rotor, the stabilizer bar prevents sudden changes of rotor-blade angle (hence tilting of the rotor disc) which can be caused by inadvertent swinging of the fuselage and mast by gusts or other disturbances.

DIMENSIONS - TAIL ROTOR

Diameter	Blade area (total) 2.40 sq. ft.
Number of blades 2	Disc area 25.31 sq. ft.

CONTROL SYSTEM

MAIN ROTOR

A conventional control stick is provided for tilting the main rotor by means of cyclic rotor blade angle change. One such stick is located in front of each seat. Either stick is removable on the ground should additional space be desired. The tilt of the rotor plane, and hence direction of flight, follows the corresponding motion of the stick. An adjustable friction device on the stick is employed to regulate control sensitivity.

A main rotor pitch control lever, operation of which causes each rotor blade to increase or decrease its geometric angle a like amount, is located at the left of each seat. Upward motion of this lever produces more lift; downward motion, less lift.

TAIL ROTOR

A pair of rudder pedals, conventionally located ahead of each seat, are connected to a pitch-changing mechanism on the tail rotor. Operation of these pedals permits both proper torque compensation and also directional control.

EQUIPMENT

POWER-PLANT

ENGINE Air-cooled motors, Model 6V4-178-B3, six cylinders opposed, unsupercharged, rated 178 horsepower at 3000 rpm. at sea-level.

TRANSMISSION This unit incorporates the following components:

- (1) 9:1 engine-to-rotor gear-reduction ratio.
- (2) Centrifugal clutch to permit starting of engine without engaging rotor.
- (3) Free-wheeling unit so that rotor may turn independent of the engine.
- (4) Auxiliary drive for cooling fan.
- (5) Auxiliary drive for tail rotor extension drive shaft.

COOLING FAN Adequate air-flow for both engine and oil cooler is provided by the fan. It is vee-belt driven from a power take-off pulley on the transmission.

ENGINE CONTROLS These include throttle, carburetor hot air, and carburetor mixture control. The throttle is located on the main rotor pitch control lever.

INSTRUMENTS

FLIGHT:

Air-speed

Altimeter

Compass

ENGINE:

Dual Tachometer (engine and rotor)

Oil Pressure (engine and transmission)

Oil Temperature Fuel Pressure Manifold Pressure

Cylinder-head Temperature

Fuel Gauge

ELECTRICAL INSTALLATION

Starter

Generator

Ammeter

12-volt, 55-ampere-hour battery

Red (port), green (starboard) and white (tail) position lights

Instrument panel light

Radio — Hallicrafter Model CA-2 transmitter and receiver

Auxiliary Fuel Pump

LANDING GEAR

Non-retractable four-wheel landing gear equipped with shimmy-dampened 360° swiveling front wheels

MISCELLANEOUS EQUIPMENT

Main rotor tie-down boot

Back and removable seat cushions

Safety belts

Tool kit

Log book

Rudder pedal locking bar

Glove compartment

Throat Microphone with extension cord

Parts catalog

Erection and Maintenance Manual

Pilot's operating handbook

Hole cover (for removable stick)

COLOR AND UPHOLSTERY

The following exterior color combinations may be selected at the customer's option:

Principal Color

Trim Color

Vermilion Salt Lake green

Cream Cream

Cream

Salt Lake green Honolulu blue

The standard model will be upholstered as follows:

Side-wallTan Vinyl-coated fabric

TrimBeige lacquer

SeatsGrey-green Bedford cord
CarpetBurnt umber Avtrim

The customer will be charged on a time-and-material basis for deviations from the standard helicopter.

NOTE: These specifications are tentative and subject to change without notice.